

Optimal Hohmann-Type Impulsive Ellipse-to-Ellipse Coplanar Rendezvous

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Abstract

This paper devotes to the problem of ellipse-to-ellipse coplanar rendezvous, where the solution and distribution of Hohmann-type optimal impulsive rendezvous are investigated. The analytical relation between the initial states and rendezvous time are derived for Hohmann-type, and the optimal impulse amplitudes are given thereupon. The distribution boundary of Hohmann-type model is obtained according to the Hohmann transfer and Hohmann with coasts. Simulations are demonstrated to analyze the influences of the solution and distribution.

Keywords: Optimal impulsive rendezvous, Hohmann-type rendezvous, ellipse-to-ellipse, optimal distribution.

1. Introduction

Optimal impulsive rendezvous is aimed at obtaining minimum-fuel guidance strategy for spacecraft rendezvous, which has attracted considerable attention. Despite that Lawden's necessary conditions¹ for optimal impulsive trajectories and Lion's improving methods² for non-optimal trajectories have provided some guidelines to solve the optimal problem where the initial states and rendezvous time are specified, the distributions of optimal models cannot be obtained clearly in these way. So far, only Prussing's theory³ of optimal impulsive rendezvous on close circular orbits is complete in its theoretical system, which derives the solutions and distributions of optimal impulsive models by solving the primer vector equations and boundary value problem. A reference frame in mean velocity orbit was built by Frank⁴, and showed better performance in describing the impulse locations and magnitudes than the mean radius orbit in Prussing's results. Xie⁵ focused on the selection of reference frame for optimal impulsive rendezvous, and investigated the effect on the

classification, distribution and guidance precision. For the case of elliptic orbit rendezvous, Wang⁶ used the state transition matrix given by Yamanaka⁷ to calculate the optimal solution of four-impulse model, but the analytical solution and the distribution are difficult to be achieved. Chen^{8,9} studied the ellipse-to-circle coplanar rendezvous based on his results on the dynamical equations for elliptic orbit rendezvous in low eccentricity, and provided the solutions and distributions of all types optimal models. Motivated by which, our previous work¹⁰ considered the ellipse-to-ellipse coplanar rendezvous and obtained the analytical solution and distribution of four-impulse model. In this paper, we will further investigate the Hohmann-type model for optimal impulsive ellipse-to-ellipse coplanar rendezvous.

2. Dynamics Description

The relative motion between two spacecrafts in elliptic orbits was derived in our previous work¹⁰, which is still used in this paper and given as follow:

$$\begin{cases} \delta\dot{r} = 3\delta r + 2\delta\theta \\ \delta\ddot{\theta} = -2\delta\dot{r} \end{cases} \quad (1)$$

The initial and terminal states of system (1) are

$$\begin{aligned} x_0 &= [x_{01}, x_{02}, x_{03}, x_{04}]^T \\ x_F &= [0, 0, 0, 0]^T \end{aligned} \quad (2)$$

where

$$\begin{aligned} x_{01} &= k_c (1 + e_c \cos f_c)^{-1} - k_t (1 + e_t \cos f_t)^{-1} \\ x_{02} &= \beta \\ x_{03} &= k_c^{-\frac{1}{2}} e_c \sin f_c - k_t^{-\frac{1}{2}} e_t \sin f_t \\ x_{04} &= k_c^{-\frac{3}{2}} (1 + e_c \cos f_c)^2 - k_t^{-\frac{3}{2}} (1 + e_t \cos f_t)^2 \end{aligned} \quad (3)$$

β is the difference of phase angle between two spacecrafts; e_c, e_t and f_c, f_t are their eccentricities and true anomalies, respectively.

The states at phase angle τ was also deduced[10]:

$$x(\tau) = \begin{bmatrix} 2d_4 - d_3 \cos(\tau + \varphi) \\ \beta - 3d_4\tau + 2d_3 \sin(\tau + \varphi) - 2d_3 \sin\varphi \\ d_3 \sin(\tau + \varphi) \\ -3d_4 + 2d_3 \cos(\tau + \varphi) \end{bmatrix} \quad (4)$$

where

$$\begin{aligned} d_1 &= x_{03}, d_2 = 3x_{01} + 2x_{04} \\ d_3 &= \sqrt{d_1^2 + d_2^2}, d_4 = 2x_{01} + x_{04}, \varphi = \arcsin\left(\frac{d_1}{d_3}\right) \end{aligned} \quad (5)$$

3. Optimal Hohmann-Type Rendezvous

The solution to primer vector equations corresponding to system (1) can be given in the following form:

$$\begin{aligned} \lambda_1 &= A \cos \tau + B \sin \tau + 2C \\ \lambda_2 &= 2B \cos \tau - 2A \sin \tau - 3C\tau + D \end{aligned} \quad (6)$$

Hohmann-type model is a special case of optimal two-impulse rendezvous, where the coefficients of (6) are

$$A = B = C = 0, D = \pm 1 \quad (7)$$

then $\lambda_1 = 0, |\lambda_2| = 1$. It can be verified that the necessary conditions of optimal impulsive rendezvous are satisfied for any phase angle τ .

3.1. Solution of Hohmann transfer

The impulse direction can be obtained from the solution (6), while the impulse time and magnitudes needed be calculated according to the following boundary value problem:

$$\begin{bmatrix} 2(1-C_\tau) & 0 \\ 4S_\tau - 3\tau_F & 0 \\ 2S_\tau & 0 \\ -3 + 4C_\tau & 1 \end{bmatrix} \begin{bmatrix} \Delta V_1 \\ \Delta V_2 \end{bmatrix} = x(\tau_F) \quad (8)$$

where $S_\tau = \sin \tau, C_\tau = \cos \tau$, ΔV_1 and ΔV_2 are the magnitudes of two impulse, and τ_F is the rendezvous time. From (4) and (8), we have

$$\begin{aligned} \Delta V_1 &= -\frac{d_3 \sin(\tau_F + \varphi)}{2S_\tau} \\ \Delta V_2 &= \frac{-2d_4 + d_3 \cos(\tau_F + \varphi)}{2(1-C_\tau)} \end{aligned} \quad (9)$$

then

$$d_3 (\sin(\tau_F + \varphi) - \sin\varphi) = 2d_4 \sin\tau_F \quad (10)$$

Substituting (5) into (10), it can be obtained that

$$\sqrt{x_{01}^2 + d_1^2} \sin(\tau_F + \theta) = -d_1 \quad (11)$$

where

$$\cos\theta = \frac{x_{01}}{\sqrt{x_{01}^2 + d_1^2}}, \sin\theta = -\frac{d_1}{\sqrt{x_{01}^2 + d_1^2}} \quad (12)$$

On the other hand, from the second row of (8), it has

$$\Delta V_1 = \frac{-\beta + 3d_4\tau - 2d_3 \sin(\tau_F + \varphi) + 2d_3 \sin\varphi}{(4S_\tau - 3\tau_F)} \quad (13)$$

Combining (5), (9) and (13), it has

$$-\frac{3\tau_F (x_{01}C_\tau + d_1S_\tau)}{2(1-C_\tau)} + 2d_1 = f_c - f_t \quad (14)$$

The appropriate initial states and rendezvous time which satisfy the necessary conditions of Hohmann transfer can be obtained by solving (11) and (14) together, and then, the second impulse ΔV_2 can be obtained as

$$\Delta V_2 = -d_4 + \frac{d_3 \sin(\tau_F + \varphi)}{2S_\tau} \quad (15)$$

3.2. Distribution of Hohmann-type model

The distribution of optimal Hohmann-type rendezvous is to illustrate the existence of feasible solution. To investigate the distribution, rendezvous time is chosen as the X-coordinate and the special phase angle defined below as Y-coordinate:

$$\delta\theta_F = -\beta + 1.5d_4\tau_F - d_3 \sin(\tau_F + \varphi) + d_3 \sin \varphi \quad (16)$$

Let τ_{Fh} be rendezvous time solved by (11) and (14), and $\delta\theta_{Fh}$ is the corresponding special phase angle. If $\tau_F = \tau_{Fh}$ and $\delta\theta_F = \delta\theta_{Fh}$, then it is just the Hohmann transfer. The two impulses are implemented at $\tau_1 = 0$ and $\tau_2 = \tau_F$. However, when the real rendezvous time is longer than τ_{Fh} , the coasts are needed to save the fuel.

If $\tau_F > \tau_{Fh}$ and $\delta\theta_F = \delta\theta_{Fh}$, it is a Hohmann model with terminal coast. The two impulses are implemented at $\tau_1 = 0$ and $\tau_2 = \tau_{Fh}$, and the residual time $\tau_F - \tau_{Fh}$ is for terminal coast. The special phase angle $\delta\theta_F$ and rendezvous time τ_F should satisfy the following relation

$$\delta\theta_F = \delta\theta_{Fh} + 1.5d_4(\tau_F - \tau_{Fh}) + d_3 \sin(\tau_{Fh} + \varphi) - d_3 \sin(\tau_F + \varphi) \quad (17)$$

$$\tau_{Fh} = \tau_2 - \tau_1, \tau_F \in [\tau_{Fh}, +\infty)$$

If $\tau_F > \tau_{Fh}$ and the special phase angle $\delta\theta_F$ satisfies

$$\delta\theta_F = \delta\theta_{Fh} - 1.5d_4(\tau_F - \tau_{Fh}) + d_3 \sin(\tau_F - \tau_{Fh} + \varphi) - d_3 \sin \varphi \quad (18)$$

$$\tau_{Fh} = \tau_2 - \tau_1, \tau_F \in [\tau_{Fh}, +\infty)$$

then, after the initial coast for time $\tau_F - \tau_{Fh}$, the special phase angle will become exactly $\delta\theta_{Fh}$. This case is a Hohmann model with initial coast, and the impulses are implemented at $\tau_1 = \tau_F - \tau_{Fh}$ and $\tau_2 = \tau_F$.

Let

$$\delta\theta_{Fh1} = \delta\theta_{Fh} + 1.5d_4(\tau_F - \tau_{Fh}) + d_3 \sin(\tau_{Fh} + \varphi) - d_3 \sin(\tau_F + \varphi) \quad (19)$$

$$\delta\theta_{Fh2} = \delta\theta_{Fh} - 1.5d_4(\tau_F - \tau_{Fh}) + d_3 \sin(\tau_F - \tau_{Fh} + \varphi) - d_3 \sin \varphi$$

If $\tau_F > \tau_{Fh}$ and $\delta\theta_{Fh1} < \delta\theta_F < \delta\theta_{Fh2}$, then there exists a Hohmann model with both initial coast and terminal coast. As shown in Fig.1., this case is illustrated in the middle of the curves expressed by (17) and (18), that is the shadow part. Denote $(\tau_{F0}, \delta\theta_{F0})$ as the intersection point of the curves determined by (16) and (18), then the two impulses are implemented at $\tau_1 = \tau_{F0} - \tau_{Fh}$ and $\tau_2 = \tau_{F0}$. The initial and terminal coast last for time $\tau_{F0} - \tau_{Fh}$ and $\tau_F - \tau_{F0}$, respectively.

From the above, the optimal Hohmann-type impulsive rendezvous has four models, all of whose impulse magnitudes are determined by (9) and (15), and impulse direction is along the tangential direction.

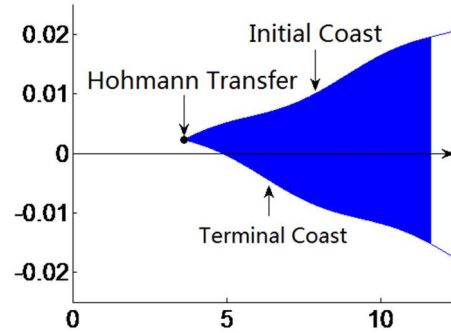


Fig 1 Distribution of Hohmann-type model

4. Simulations

In this section, simulation examples are presented to show the guidance performance and distribution of Hohamman-type impulsive rendezvous.

4.1. Hohmann ellipse-to-ellipse rendezvous

It is assumed the semi-major axis and eccentricities of the target orbit and chaser orbit are initially $a_t = 6730$ (km), $a_c = 6750$ (km), $e_t = 0.0005$ and $e_c = 0.0004$, respectively. Let τ_F and β (rad) be the appropriate rendezvous time and initial difference of phase angle, respectively, which satisfy (11) and (14). And denote R_r (m) as the optimal radius of reference frame, R (m) as the initial relative distance, Δa (m), Δe and $\Delta\theta$ (rad) as the guidance errors.

Table 1. Results of Hohmann impulsive rendezvous

	1	2	3	4
f_c	30°	150°	210°	330°
τ_F	3.62	3.44	2.80	2.68
β	7.12e-03	9.68e-03	6.04e-03	3.46e-03
R_r	6.70e+06	7.92e+06	7.81e+06	6.71e+06
R	4.99e+04	7.05e+04	4.83e+04	2.71e+04
Δa	6.80e+01	3.38e+04	3.09e+04	2.49e+01
Δe	3.38e-05	4.51e-03	4.22e-03	8.92e-06
$\Delta\theta$	5.82e-06	2.87e-04	3.50e-05	3.80e-06
ΔR	1.29e+02	5.14e+03	5.71e+02	5.08e+01

Simulation results of Hohmann transfer for ellipse-to-ellipse rendezvous are demonstrated in Table 1, which shows that: (1) with different true anomalies, even if the other initial states are the same, the

rendezvous time and initial difference of phase angle which satisfy (11) and (14) varies much; (2) the optimal radius of reference frame also changes with the true anomaly; (3) the guidance precision is high when the chaser initially stays around the perigee.

4.2. Distribution of Hohmann-type model

To investigate the distribution of Hohmann-type ellipse-to-ellipse rendezvous, we take rendezvous time τ_F as the X-coordinate and $\delta\theta_F/d_4$ as Y-coordinate. Fig.2 shows the distribution of Hohmann-type model with different true anomalies and eccentricities.

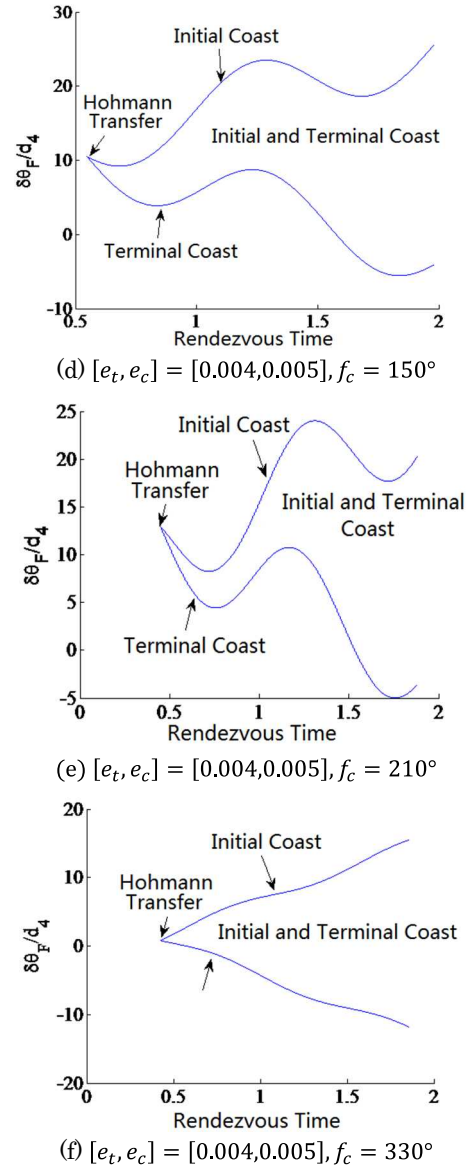
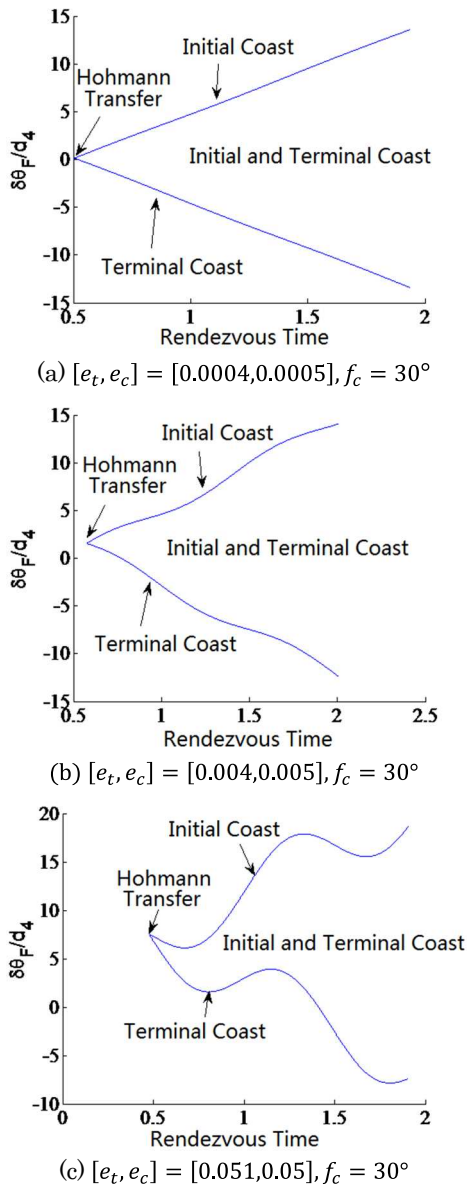


Fig 2 Distributions with different true anomalies and eccentricities

5. Conclusion

This paper extends our previous work¹⁰ to the Hohmann-type optimal impulsive rendezvous. By defining the special phase angle, we derived the analytical solution for Hohmann transfer, and obtained that the optimal Hohmann-type impulsive rendezvous has four models, i.e. Hohmann transfer, Hohmann with initial coast, Hohmann with terminal coast and Hohmann with both coasts. In further research, we will integrate all optimal models in one map, including four-

impulse, three-impulse, three-impulse with coasts, two-impulse, two-impulse with coasts, and Hohmann-type.

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